

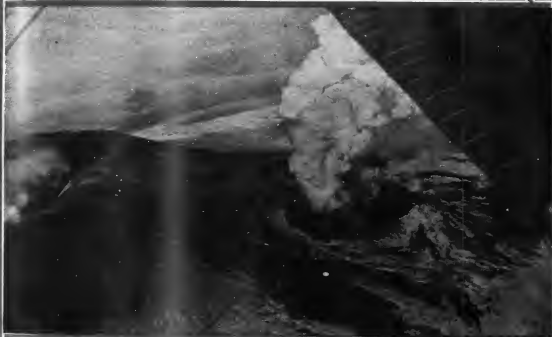
# AVIATION

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JUNE 15, 1925

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Northeast crater of Mount Etna in activity

VOLUME  
XVIII

## SPECIAL FEATURES

NUMBER  
24

LOGAN FIELD CROSS-COUNTRY

THE DESIGN OF AIR COOLED CYLINDERS

GENERAL PATRICK ON THE INDEPENDENT AIR FORCE

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# AVIATION

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### Real Leadership

**B**UILDING in the several thousand pages of testimony taken before The Leapfrog Committee is a document that ought to be kept before everyone who has the future welfare of the Army Air Service at heart. It is the recommendation of General Patrick to the War Department for the reorganization of the Air Service and it was given to the Committee by the Secretary of War. This short summary cannot cover sound thought and correct principles that any such document that has come to the notice of Americans in the last ten years. It is printed in full elsewhere in this issue and it did not only be read, but made the program of everyone who intends to make suggestions to Congress for constructive legislation for the Air Service during the next session.

Summarized, General Patrick's plan calls for the following changes:

- 1. Creating a united air force.
- 2. Immediately, a separate corps.
- 3. A single supreme air commandant.
- 4. Air Force Defense concentrated.
- 5. Separate tactical training.
- 6. Separate Budget, and Estimates.
- 7. Single command.

- 8. Separate procurement policy.
- 9. Separate supply and distribution policy.

A careful reading of the presentation of the above suggestions will confirm the impression that the recommendations are as sound as their reasoning and as practical as their objectives as have been made by anyone during the controversy that has raged over the whole question of the future of aerial air defense. It took courage of the highest order for an officer with a distinguished Army career leaving the land of retirement to propose a line of action which would take all studies of aviation away from the General Staff.

If we offer to the Navy would formulate a similar debate and independent plan for the reorganization of naval aviation and then some other well informed person, with influence in official circles would propose a policy for the reorganization of civil aviation by the government, perhaps those who have been discouraged since the war by the steady decline of trend of government aeronautics would find that the efforts that have been made by everyone who has not completely succumbed to the belief for official Navy have not been fruitless.

It is most significant that an officer with the high reputation of General Patrick should come out frankly and tell the General Staff that the Air Service would function better without its control. Since he has been in command of the Air Force, Americans have had occasion to differ often with him, as the policies of the Chief of Air Service, but this statement is so completely in accord with the ideas that have been so frequently expressed in these columns that we cannot but

find that if the same conclusions can be reached from an entirely different and sound, more reasonable chain of premises, that great hope can be entertained that Congress will take some action that will correct a mistake that is holding back the development of our military aviation.

Acting Secretary of War Dene has stated that General Patrick's recommendation "is probably the best statement of the needs of the country from the standpoint of the Air Service that we have." At this time Americans wish to go on record that it will support every one of the proposals made by the General and furthermore commendation has on having the leadership that will lead to a back of followers. May we make a suggestion to those who believe that General Patrick has stated a sound and fundamental policy. Just write him a letter and give him the benefit of your suggestions and encouragement. Only by such encouragement can he be expected to lead the movement for a national military air policy.

### The New Air Mail Head

**W**HILE the regret is general over the resignation of Col. Paul Henderson as head of the Air Mail, the sense of his successor indicates that there will be no change of policy in the conduct of the outstanding triumph of American Aviation. It is a unique example on governmental records that may be considered as a small part of the responsibility of one of the numerous important national positions should involve having face and credit to the men who have with voice and courage seen the Air Mail through to its present world famous status. The names of Frazier and Henderson as well as that of Ragsdale who was last in the position—leader of the service, will be indelibly written in sharp and lasting letters on the tablets of the history of the progress of aviation. Other, it is overlooked that the Second Assistant Postmaster General has charge of all the transportation of mail—railway and steamship, as well as by airplane. The only difference is that the Air Mail is operated by the Post Office Department in the same way automobile trucks are directly under the control of the Postmaster General.

The new Air Mail head, the Honorable W. Irving Glover has been. Third Assistant Postmaster General for four years. He has made a splendid record in handling the very complicated relations of publications with the Government so they have to do with postal rates. His new appointment by the President is an indication of the high regard in which he is held in official circles in Washington. Taking charge of the Air Mail at one of the most critical times in its history, his opportunity will be even greater than his predecessors. Every one connected with aeronautics will wish him success and hope that he may enter into the new field with the assistance of good will and advice that has always been one of the pleasant features of the office.



# The Design of Air-Cooled Cylinders

By C. FAYETTE TAYLOR

The cooling of the cylinder barrel below the combustion chamber is largely a problem of proper cooling design. The remarks on the design which follow are applicable not only to the cylinder barrel, but to the combustion chamber walls and ports.

The arrangement of fins on an air-cooled cylinder depends primarily on the position of the cylinder with respect to the air blast. For approximate work, it is now customary to assume that the air blast is directed against the side of the cylinder at approximately right angles to the cylinder axis. When this is the case, some type of circumferential finning is required. Circumferential finning has the following advantages as compared with flange parallel to the cylinder axis:

- (a) Adds greatly to the strength of the cylinder to withstand internal pressure.
- (b) Is easily machined where machining is necessary.
- (c) Is much more in cost than the axial finning.
- (d) Gives more cooling surface on the ports and combustion chamber.

The question of the proper size and spacing of the fins in one which has received a great deal of attention and has been the subject of a number of extensive research programs. The size and spacing of fins is largely controlled by the conditions of the material used. However, for most materials, the circumferentially spaced finning will be the most efficient together and at such this section that the manufacturing difficulties become prohibitive. Another consideration is that the fins must be sufficiently thick and strong to resist damage in handling. The finning to be used, therefore, is the result of a compromise between these various factors. It has been found that for aluminum or cast iron having a very good design from all points of view is to make the fins 1/8 in. long with a thickness of 0.025 in. at the root and 0.0625 in. at the tip, with a height of 0.375 in. between fins. For steel, the work is best done with a depth of fin of 0.5 in. For the fin spacing is usually increased where the finning interferes with the valve ports, so otherwise the manufacturing problems are greatly reduced. In general, the fin arrangement must allow on a number of several around air-cooled cylinders.

Considering the material of the cylinder from the point of view of cooling, the following are the factors to be considered:

1. material having a high rate of heat transfer from its surface to air
2. a material having rapid internal conductivity, so that the heat will flow freely from the lower to the outer parts of the cylinder, thus avoiding local overheating.

No definite data on the rate of heat transfer to air is available, but from experience it appears that stainless, steel and cast iron possess this property to a high degree.

## Aluminum

Of the metals having high thermal conductivity, aluminum stands out as second to the added advantage of light weight, although copper and its alloys have been proposed and used. The question of cylinder materials, however, is largely governed by considerations other than those of cooling and will be treated at greater length under the next heading.

As previously stated, durability is stressed largely by the proper selection and treatment of materials, and the proper design and lubrication of bearing surfaces. Adequate methods, which has already been discussed, also promotes durability.

The selection of materials for air-cooled cylinders, or for any other type of engine, is governed not only by the question of durability, but also by questions of weight and methods of manufacture. Special considerations, such as resistance to corrosion and appearance, may also affect the choice of material.

Finally, however, as durability is usually the primary factor, the selection of materials will be considered at length at the end.

In choosing the material for any particular part, it is first necessary to decide whether the part is primarily a structural or bearing member, a reinforcing member, an air duct or liquid, a bearing member, or, as is usually the case, a member combining two or more of these functions.

## Cylinder Stresses

Let us consider, for example, the cylinder body proper. This particular member combines all of the functions just mentioned. It is a structural member subject to high stresses due to gas pressure, forces at the full-down flange, side thrust of the piston, and forces imposed by the operation of the valves and valve gear. It is a bearing member where the piston and piston rings slide in the bore, and it is a reinforcing member for the gross compression and for lubrication. Furthermore, it is a part of complicated shape, which must be made by casting, making its manufacturing cost in the process. Cost, then, will fill all of the functions in a bearing member, but its weight is generally very low, at least for military engines. For industrial engines and non-cylinders may often be used to advantage. Cast aluminum alloy is satisfactory for a medium size design and can be cast in intricate shapes. It will not stand a high degree of stress, and it is a fairly good bearing material. However, where high loaded stresses are encountered, cast aluminum alloys are liable to fatigue by design and consequently have been found unsatisfactory for the full-down flange, where the cylinder is held to the crankcase under very heavy stresses are met. Although aluminum is a good bearing material under moderate stresses, and is satisfactory for the bearing surface of the piston. This is largely because the piston itself must be of aluminum and two materials of the same kind can be cast in one piece. For the full-down flange, a flat cast iron bearing surface. Furthermore, carbon particles from the combustion chamber will wear the comparatively soft aluminum of the cylinder walls. A few engines have been cast in steel, but this is not a satisfactory material for the cylinder bore, and in these it has been found that after long periods of running the bore shows rather deep wear and scoring. There long life and extreme durability are desired, therefore, it is evident that the aluminum alloy is present available is not a satisfactory material.

In order to overcome these difficulties, a combination of materials has usually been resorted to. Steel rings in sufficient bearing surface for the piston, and at the full-down flange, a flat cast iron bearing surface. Furthermore, carbon particles from the combustion chamber will wear the comparatively soft aluminum of the cylinder walls. A few engines have been cast in steel, but this is not a satisfactory material for the cylinder bore, and in these it has been found that after long periods of running the bore shows rather deep wear and scoring. There long life and extreme durability are desired, therefore, it is evident that the aluminum alloy is present available is not a satisfactory material.

Another alternative for the material of the cylinder body is to make it entirely of steel, as was done in the case of the rotating cylinder, or "rotary" engines, which were popular

for a few years ago. Steel is a satisfactory material from the point of view of weight, since its strength makes possible the use of very light sections. It is also one of the best structural materials. It is satisfactory as a reinforcing member and appears to be satisfactory from the point of view of casting. The greatest objection to the all-steel cylinder is the expense of manufacture. It has not been found satisfactory in the cylinder as a bearing member, and it is therefore necessary to machine the cylinder from a forging, an extremely long, difficult and expensive process. On this account the all-steel cylinder has been practically abandoned for non-critical use.

## Valve Ports

The valve ports are primarily gas containing members, although they are subject to more stresses imposed by the main body and valve gear. When the cylinder is made from a material it is best to make the valve ports of the same material. This is quite satisfactory from this point of view.

The issue of material for the valve ports is comparatively easy. It is a gas containing member subject to fairly high stress and shock loading, with a bearing surface along the axis. In ordinary high-grade alloy steel will be satisfactory for the purpose, provided the valve seats can be well taken care of. If the valve seats are made of a harder material, such as cast iron or "inlay" valve seats, it may be necessary to make the valve valve from an extremely hard material, such as a case hardened or a hardened alloy steel.

The conditions of operation of the valve ports are like those of the intake valve, except that the exhaust valve must operate at extremely high temperatures. For this part it is necessary to use a material which has a high strength at high temperatures, and is not affected by a corrosive action of hot exhaust gases. Special steels containing boron, chromium or silicon as the principal alloying elements are necessary for this part. For a more complete discussion of this subject the reader is referred to E. D. Hesse's article in the *ASA Bulletin* for August, 1934.

The valve ports are essentially bearing members, and the material to be selected depends largely on the amount of wear to be expected. In the bearing surface of the valve ports, the "inlay" type of valve port where there is little or no lubrication, the grades should be made of extremely hard material, such as case hardened or tungsten steel. The latter is usually specified for the exhaust valve grade, as it will retain a large measure of the hardness at high temperatures. Where the valve grade is well lubricated, or with an oil-cooled and properly sized valve gear, the material may be used on the intake valve ports, but it is also desirable to use hard material for the exhaust grade on account of the high temperature at which the intake and of the gas to be used to operate it. The subject of valve grades is covered extensively in the article by E. D. Hesse previously mentioned.

## Alloys

The consideration in detail of the various alloys of steel, aluminum and copper (brasses) is outside the scope of this article, as it is well known, however, that the metal program as the development of non-ferrous alloys has been due in no small degree to the fundamental progress in the treatment of metallurgy and metallurgy which has taken place in the last few years. This is especially true in the field of aluminum alloys, which have been developed to a point where they are now being used in the design of aircraft. Table 2 gives the principal characteristics of a number of the more important aluminum alloys used in the aircraft industry. The alloys are listed into every phase of design, but will be treated here from the point of view of bearing satisfactory durability. In designing any part, it is necessary to analyze the stresses and strains to which the part is subjected. The material to be used must be classified roughly as

direct stresses, vibrational stresses, stresses imposed in handling, wear by friction, and corrosion.

Of all the deteriorating factors mentioned, the direct stresses are the only ones which can be analyzed quantitatively with any accuracy. The question of the effects of direct stresses are very valuable when their limitations are definitely laid to rest. In the case of the cylinder, it is possible to calculate fairly closely the stresses caused by the pressure of the gases against the cylinder walls and cylinder head, and the resultant stresses imposed on the flange where the cylinder is bolted to the crankcase. Other simple stresses, such as the side thrust of the piston against the cylinder wall and the stress in the joint between the cylinder head and cylinder head, can also be calculated with a fair degree of precision. From these calculations it is possible to determine the amount of material to be used and these points are being working efforts for cylinder stresses, which can be calculated, such as those due to vibration and handling. In order to take into account these unknown stresses, it is customary to introduce what is known as a "factor of safety." After determining the maximum stresses of material to be withstood the direct stresses, the result is multiplied by the factor of safety in order to obtain an ample margin to cover the unknown stresses.

## Factor of Safety

The question of factors of safety is intimately related to the question of fatigue of materials. It is known that metal parts subject to vibration or to frequently repeated loads or shocks may eventually fail, even though the stresses imposed by these loads are well below the yield point of the material. In such cases the material will stand under a stationary or gradually applied load. A large number of experimental investigations have been made to determine the effect of vibration on various materials. The fatigue limit is defined as the maximum stress which a given material will stand indefinitely under conditions where the load is rapidly applied and removed or the direction of loading rapidly reversed, as in the case of parts subject to severe vibration or to alternating stresses. Naturally, the fatigue limit is considerably lower than the ultimate strength of a material, and therefore the factor of safety must be increased when the material is subject to this type of loading. Almost all of the parts of a cylinder are subject to vibration, and most of them are rapidly changed in the operation of the engine. The material of the cylinder should, therefore, be chosen in terms of its fatigue limit. The subject of fatigue of materials cannot be treated at length in this article, but it is one which is of such importance in the design of aircraft that it is worth a few words on this subject have been used in recent years.

The temperature at which the various parts will operate is an especially important consideration in air-cooled cylinders, where the temperature of the cylinder walls and cylinder head is much higher than in usual for ordinary engine parts. The strength of metals decreases rapidly with increasing temperature, and it is therefore necessary to allow larger sections of material to be used at high temperatures. The factor of safety should be increased when the temperature of operation is high. Factors of safety should be based on the characteristics of the material at the operating temperature.

Another consideration in determining the factor of safety is the amount of deflection allowable. On most engine parts, including the cylinder, very little deflection is permissible and parts must be designed with this consideration in mind.

In designing bearing surfaces, where ample lubrication can be provided, it is best to form the bearing of two different metals, one hard, such as steel, and one soft, such as brass or aluminum. Where little or no lubrication is available, the bearing should be made as hard as possible, so as to reduce the wear to a minimum. This has already been mentioned in the case of valve seats and valve guides. The larger the bearing surface, the more the bearing will be subjected to the stresses and the greater will be the durability of the bearing.

gylinder design, however, the available bearing surface is usually limited by space restrictions. It has been found best to allow for a total bearing length on the piston of at least two-thirds its diameter. Valve guides are subject to much higher unit pressures and, if possible, their length should be at least five times their diameter, and the diameter should be as great as possible without unduly restricting the valve ports by large valve stems. For the exhaust valve, a stem diameter of one-quarter to one-half the valve diameter has been found to give the best results. For intake valves the stem may be made somewhat smaller.

### News@11pm

The question of allowance for handling of the parts comes up particularly in the case of the cooling fins, which can be easily broken if the cylinder is not carefully handled. It has been found that the six dimensions suggested earlier on this article are sufficient to withstand reasonable handling, but that the fins are made much thinner, they will frequently break. The cooling fins are not the only parts of the engine parts where handling must be considered as those with threads, such as bolts and studs, where the stresses imposed on threading may be higher than the stresses caused by engine operation. It has been found by experience that dry soapstone is a good material for the threads of bolts and studs. It should never be used, and a modulus of 0.0025 to 1 is better to refer to good stripping of the threads by carbon compounds.

Parts subject to the corrosive action of hot exhaust gases must be made with larger sections than of the exhaust valve and piston, since corrosion of the first gradually reduces the amount of metal available to fulfill the other functions of the valve. The same is true of the piston rings, which support and cylinder head. The last method of solving this problem is to choose materials which have the maximum possible resistance to corrosion. For exhaust valves, steels containing chrome and chromium as the principal alloying elements have been found very satisfactory in this respect, and the same is true of the piston rings. The valves themselves do not, however, free from action of this nature.

The problems of distribution of materials cannot be reduced to hard and fast rules which will give the necessary thickness of material for any part. The actual size and arrangement of any part are, after all, dictated largely by experience, which must be obtained by gradually working up under actual operating conditions and study of previous successful designs. Without this experience the best designers could not hope to produce a successful estimator.

### Embryonation

[illegible]

The weight of the parts which go to make up an aircraft engine must be completely kept in mind by the designer. Every pound which is put into the engine means a sacrifice of an equivalent weight which would otherwise be available.

for useful load. The low weight is proportion to the power output which is possible in internal combustion engines is largely responsible for the present state of aeronautical development. The introduction of alloys of aluminum has assisted greatly in weight reduction, although the increase in the quality and tensile strength of alloy steels has loss of almost equal assistance. Table 4 shows the weight per unit of volume and the tensile strength of some of the common alloys used in air-cooled cylinder construction.

While the choice of material is important in achieving the maximum weight, the elimination of unnecessary material is equally important, and the design should therefore be based in every detail to be sure that no material is produced which does not perform a very definite function. The thickness of all sections should be held to the minimum consistent with strength and durability. There again much improvement is of great importance in determining how the use can be made of the sections of the various parts before they are rejected as flawed.

*M. agrestis*

Recently, alloys containing a large percentage of zirconium have been proposed for use in nuclear reactor engines. While these materials are still in the experimental stage and cannot be considered completely satisfactory except for glaucometer use, they are not subject to high stresses. In fact, the writer believes that alloys of this type will eventually be used and will offer a further reduction in the weight of reaction alloys. An alloy of this nature has been evaluated in tables 2 and 4 for comparison with steel and aluminum alloys now commonly used.

By careful design it is possible to avoid secondary manufacturing costs without sacrificing any of the desirable characteristics usually implied. In general, it may be said that the most economical design is that which is most easily and most accurately assembled. It is possible to design a machine so arranged as much as possible by grinding, forging or stamping, if production costs are to be kept to a minimum. On the other hand, the operator's life is *short* when machines are so arranged that the operator has to do a great deal of work by turning the various parts of the parts, with consequent increase in weight. Since that is not usually permissible in assembly, the most economical design is that which is most easily assembled. Thus being the case, the machine work should be confined to such as is possible to the cheaper operations. Expensive operations, such as grinding, and milling, and turning, should be confined to the parts which are of great depth, should be avoided. Here again the danger must be guarded on the experience and knowledge of shop men, and the designer must be careful to select those who will have the benefit of the manufacturer of the parts.

### Conclusions

The field of advanced cylinder development is still in its infancy, and it is dangerous to make definite predictions for its future. If an engine may be constructed, it is in the effort that the basic principles set forth have been definitely established, and that future progress will be confined to detail refinements as long as the outboard four-stroke cycle is retained. The greatest fields for detail improvement appear to be in reducing weight by the use of materials having a higher strength-to-weight ratio, reducing weight by increasing the use of aluminum alloy, and reducing weight by, in the case of the exhaust valve, an exhaust valve guide, and in the simplification of design to reduce the cost of production.

TABLE I  
Average power per cubic inch piston displacement for  
air-cooled aviation engine cylinders, 8.2 compression ratio.

Size (mm)	Brachycephalus sp. nov. (n = 10)	Brachycephalus sp. nov. (n = 10)
1.000	1.00	0.54
1.500	3.00	0.84
2.000	4.00	0.78
2.500	4.00	0.57
3.000	1.00	0.20
3.500	1.00	0.00
4.000	1.00	0.00

June 15, 1932

AVIATION

Test Treatment	Steady Strength lb in. in.	Impact lb in. in.	Flexural Strength lb in. in.	Resilience	Remarks
Control	10,000 to 15,000 lb in.	0.4	15 lb in.		Control
Asphalt	10,000 to 15,000 lb in.	0.4	15 lb in.		Asphalt
Concrete	10,000 to 15,000 lb in.	0.4	15 lb in.		Concrete
Steel	10,000 to 15,000 lb in.	0.4	15 lb in.		Steel
Aluminum	10,000 to 15,000 lb in.	0.4	15 lb in.		Aluminum
Wood	10,000 to 15,000 lb in.	0.4	15 lb in.		Wood
Plastic	10,000 to 15,000 lb in.	0.4	15 lb in.		Plastic
Other	10,000 to 15,000 lb in.	0.4	15 lb in.		Other

TABLE 6  
Factors of Risk

Material	Effect on process	Effect on maintenance	Recommended Factor of safety based on ultimate strength of steel
Steel	Tension due to pressure on cylinder head	Relatively varying load Uniform	Factor of safety 1.5
Steel	Tension due to pressure on cylinder head	Factor high momentary Variable varying load	1.5
Aluminum alloy	Tension due to pressure on cylinder head	Steady varying stresses Relatively varying load	1.5

TABLE 4  
Tide Stages through Which Birds of Various Species

[illegible]

### Plane Maturing for Intercity Airplane Tours

Arrangements to go forward with the promotion of a literary airplane reliability tour this summer were complete at a meeting of the Detroit Aviation Society on May 29 at William B. Mayo, president of the society, said a report by Selby F. Clarke, general manager of the Society of Automobile Engineers, at a meeting in Detroit the first week in June is complete on the preparation of details plan for the arrangements to be made for the conduct of the event, which will be the first of the kind.

Before leaving for the West, Mr. Clarkson issued a call for a meeting of the Aeronautical Advisory Committee of the Society of Automotive Engineers to consider the draft outline for the contest as prepared by the Contest Committee of the National Aeronautic Association.

The National Automobile Association is to be organized and associated at previous meeting in the town of the Federal Aviation Society with representative to Society of Automotive Engineers and as endorsed in 1934 by the Board of Governors of the National Automobile Association, is to start at Ford Airport, Dearborn, Mich. on Labor Day, Sept. 7, and continue for two week, with weekly stop in each of half a dozen Middle Western cities where there are good landing fields and other necessary facilities. The route will include, in order, St. Louis, Indianapolis, Evans, Columbus (Ohio) and Cleveland, each with a 10 day stop.

Members of the R.A.E. Astronautics Advisory Committee called together to consider the draft of rules for the contest.

### Some Problems on the Lift and Rolling Moment of Airplane Wings

N.A.C.A. Report No. 300

This paper is part of the thesis submitted by James B. Southworth to Johns Hopkins University for the degree of doctor of philosophy.

This report deals for the most part with the application of the aerial and treated wing theory to the calculation of the lift and pitching moment of airplane wings. Most of the results arrived at are strictly true only for wings of elliptical plan form. The investigation aims to give some indications of the accuracy with which the results can be applied to the wing forms in actual use.

# National Guard Cross Country Meet

By MAJ. W. D. TIPTON

Commanding Officer, 296 Division Air Service, ME. N. G.

A very interesting contest, unique in the history of semi-annual events, was staged Wednesday, May 30, at Logan Field by the 296 Division Air Service, the Maryland National Guard. The competition was known as the Second National Flight for a prize of \$1000 offered by the Baltimore American.

The contest took the form of a scheduled cross-country flight from Baltimore across the Chesapeake Bay to the several shore cities of Annapolis, Pocomoke, Chincoteague, and Centerville, stops being made in order named and return across the Bay to Logan Field. The flight was scheduled at the rate of one mile a minute. Each contestant was required to land on scheduled times, take off on unscheduled times, and was required to land five minutes after being sighted at each field, or otherwise suffered a penalty. Three aces were charged against the contestants whether they were early or late in either landing or taking off.

## Controls

Contestants, leaving Logan Field, were each handed an order giving the time of departure from Logan Field and the time of arrival at the next stop, Pocomoke, Md., where the contestants arrived at Pocomoke, they were given further orders to fly on unscheduled time to the next stop and so on. Observers were stationed at all air fields with orders to check to a fraction of a minute the time each contestant was sighted, the time he landed and the time he took off. Upon landing at each of the different control fields, the contestant was allowed 15 minutes to get his own orders, figure out his course and get off. At Pocomoke, the next point, the course, 45 miles, were allowed to unfold. The provision that the time of minutes as variation from the schedule whether late or early.

The idea behind the flight was two-fold, first to demonstrate the degree of training and efficiency in cross-country flying attained by officers of the local National Guard unit, and second, to demonstrate that Chesapeake Bay, while a natural barrier to surface transportation, was surmounted by the airplane and that airplanes could maintain a schedule even in early winter days without difficulty. Eight planes of the National Guard that were used and the slight times comparing were made up of six officers and one enlisted man.

Prizes offered by the Baltimore American were:

First \$500 for officer, \$100 for enlisted man.

Second \$250 for officer, \$100 for enlisted man.

Third \$100 for officer, \$50 for enlisted man.

The contest started at 9 o'clock, with Capt. Chas. A. Mason, leaving, followed at 15 min. interval by Capt. W. D. Tipton, Capt. Thos. B. Brown, Capt. Robert C. Hays, Capt. W. D. Tipton, Lt. Robinson, Lieut. James C. Hays, Lieut. Leo M. Williams and Lieut. Otto M. Hays. The day was very poor for flying but excellent for the contest. Immediately upon leaving Logan Field all pilots took to the air. It was 10 o'clock when the day and had to make the first leg to Pocomoke by sunset. Throughout the entire flight weather conditions were most the best, improving somewhat during the day.

## Winner

The winner of the contest was Capt. Harold B. Bokeman, who made the entire course with the remarkably small variation from schedule time. His scheduled time was one by Lieut. W. D. Hays, whose error was only 2.75 minutes less than the winner and third place by Lieut. James C. Hays. The winner was 10 min. off of schedule time. Lieut. Thos. B. Brown and Capt. Robert C. Hays were fourth and fifth. Three places were eliminated. Lieutenant Brown tipped over in a ditch after a flat landing at Pocomoke and crashed his propeller. Lieutenant Bokeman after blowing out both of his tires at Pocomoke, landing at Pocomoke, had been seen,

was in grief in the rather soft field at Pocomoke and turned over in his back. Lieutenant Williams suffered the only unfortunate incident of the day, a slight gasoline streamer, near Pocomoke. All ships returned to Logan Field on Sunday morning by air, after entire repairs had been effected.

All pilots will appreciate how difficult it is to land easily on the outside. The time of landing and take off in the contest were taken for landings when the wheels touched the ground and for take off when the throttle was open. For the purpose of the contest, the only exception was Lieut. Robinson and Snyder, who landed on the outside, a landing at down in detail.

	Actual Time	Scheduled Time	Delta
Harold B. Bokeman	1:10:00	1:10:00	0:00:00
Thos. B. Brown	1:10:00	1:10:00	0:00:00
Robert C. Hays	1:10:00	1:10:00	0:00:00
James C. Hays	1:10:00	1:10:00	0:00:00
Leo M. Williams	1:10:00	1:10:00	0:00:00
Otto M. Hays	1:10:00	1:10:00	0:00:00
W. D. Tipton	1:10:00	1:10:00	0:00:00
W. D. Tipton	1:10:00	1:10:00	0:00:00
W. D. Tipton	1:10:00	1:10:00	0:00:00
W. D. Tipton	1:10:00	1:10:00	0:00:00

Lieutenant Snyder varied but slightly with Lieutenant Bokeman. He made an error in taking off at one-half minute because one of his watches was not exactly synchronized with those of the others, and it showed a time variation from the schedule greater than a minute.

## Conclusions

Many interesting points are brought up from this contest. First, the two who made the best score made very small percentage for the error. They had no time to lose and the error would be, they did know the error to be voided and they had very possible means placed on the way. Further, each pilot had carefully arranged his compass and had a chart of directions. Two pilots were late in the time by Logan to Pocomoke. The answer is very simple. Although they had compasses in the back and at the front, they did not read them by the time they were in the air. They could not read the compass and neither pilot who came in late after the race had failed to check his compass and position. These three pilots made careful preparations for the contest and at the same time were not prepared. The contest was the third was eliminated by an accident.

In scheduling the course at 60 mi./hr. for ships which could reach 80, an effort was made to eliminate everything but the human elements in other words, every ship could reach the course in time. It was found that the course was not too long. The whole contest ran very smoothly. Landing fields had been selected over the shore and had been photographed in relation to the city itself. Representatives of the contest were met with equipment within one at each field only after meeting and had and a "2" shaped marker.

At the Pocomoke field a plentiful supply of gasoline was available. Excepting one forced landing due to a slight gasoline leak, every pilot's performance, good, bad, or indifferent, could be charged to the pilot himself and it is hoped that contestants of this nature are a most profitable means of checking cross-country ability of pilots, particularly class of participants for flying the course are left to the discretion of the pilot himself.

It is believed that such a contest winners all of the value at Coast Units in the Atlantic Seaboard, Maryland, Pennsylvania, New York, Massachusetts and Connecticut, could participate, would be most interesting and instructive.

# W. I. Glover to be new Air Mail Chief

Post-Office Coughlin has announced the appointment of William W. Irving Glover, Third Assistant Postmaster General, Post Office Department, to be Second Assistant Postmaster General to take effect August 1, 1933.

As the second Assistant Postmaster General has charge of all transportation of the mails, the Air Mail comes under his direction. The reason for the change is because of the resignation of Col. Paul Henderson to become General Manager of the National Air Transport which has recently been announced.

Mr. Glover was born in Brooklyn, N. Y., on Oct. 2, 1878. He was educated in the New York public schools, attending Old City graduating there, as well as in the High School. When sixteen years of age he entered the United States Army and was employed in the Old Commerce House of James T. Ford for a period of years and while there took a technical course in signaling and working at the Phoenix Wireless Company, Buffalo, Conn., becoming Junior New York Sales Representatives of the Company.

## Early Business

In 1907 organized the Adair Holding Corporation and, in 1910, the Boston Holding Corporation, both of which have been engaged in building construction in New York City and vicinity. Mr. Glover is connected with the Boston Holding Company as its Treasurer and the office of the Company are located at No. 208 Madison Avenue, New York City.

Mr. Glover entered politics early in life and his first active campaign was in behalf of Theodore Roosevelt when he ran for Governor. For years Mr. Glover was a member of the United States Republican Club. In 1910 he ran for Congress in New York, N. Y., but his name made no head. It was not long after coming to New Jersey that he took an active interest in politics and was soon elected to the Bergen County Board of Freeholders when that office was elected by Democrats and he was elected from his home city of Englewood in the County. In 1917 he was elected to the New Jersey Assembly and served for five terms, which up to that time, and was one of the most active men ever served by an assemblyman from Bergen County.

In 1920, Mr. Glover was the mainstay of the House for Republican and served with credit and distinction both as the Speaker of the House and as the Speaker of the Senate. The House of Assembly passed the 1920 Amendment after a most bitter fight and contest on the floor. The House was divided, not as political parties, but Republicans and Democrats, and Mr. Glover was the only one who was elected. The Speaker was for the passing of the 1920 Amendment and after a 544 hour debate, the resolution was passed. It was not that such parliamentary questions were presented by the Speaker but that the House was divided and Mr. Glover was the only one who was elected. The Speaker was for the passing of the 1920 Amendment and after a 544 hour debate, the resolution was passed. It was not that such parliamentary questions were presented by the Speaker but that the House was divided and Mr. Glover was the only one who was elected. The Speaker was for the passing of the 1920 Amendment and after a 544 hour debate, the resolution was passed. It was not that such parliamentary questions were presented by the Speaker but that the House was divided and Mr. Glover was the only one who was elected.

## Harding's Friend

In 1920, after the close of the Presidential campaign, in which he had taken a most active part in his home State, Mr. Glover went to Canada, America and the Panama Canal, and he spent the time that had been his home in the United States the then Senator Harding, and an organization was made with President Harding which continued until the President died.

On May, 1923, Mr. Glover was selected as the third assistant postmaster general of the Post Office Department and has since under three postmaster general—Hon. Wm. Hays, Dr. Herbert Warren and Postmaster General Harry B. New, and was elected to Civilian Work as a Member of the United States National Conference Committee where met in Ottawa, Canada, in 1923, and was later selected by the Late President Harding and Postmaster General New to represent the United States Government at the International Philatelic Congress which was held at Montreal, Quebec, Canada, in 1923. Mr. Glover was also chosen by Postmaster General

and New to represent the United States Post Office Department as Head of the Postal Commission which visited Canada in January, 1923, and served as Chairman of said Commission.



W. Irving Glover

Mr. Glover has served as Third Assistant Postmaster General for four years and during this period he has attended State Postal Conventions in every State in the Union, so doing has formed a wide acquaintance with the personnel of the Postal Service. He has greatly improved the service of the Postal Service through the medium of personal contact and the vast amount of correspondence which he has received as a result of his attendance at these conventions proves beyond a doubt that they have been of inestimable value to the Government and Postal Service as well.

During Mr. Glover's incumbency as Third Assistant Postmaster General he has brought about many improvements in the service of the Bureau. Among the most noteworthy improvements made by him has been the improvement of the postal service which has been inaugurated since his appointment as the following:

In October, 1921, he established the Philatelic Agency of the Post Office Department and has since the sales of stamps and postage stamps in every State in the Union. The total sales of the Agency, since its establishment in 1921 to the present time, amount to approximately \$200,000, which tends to prove how greatly it is appreciated by the philatelists throughout the country.

Mr. Glover also instituted a method of selling foreign money orders to the patrons of the Post Office Department which meant the saving of thousands of dollars to the patrons of our foreign money order system. The principal feature of this method was the provision of rapid changes in the rate of exchange to all of the money order offices throughout the country, thereby giving the purchaser of the money order the advantage of the change in the rate of exchange of foreign money.

Mr. Glover married Miss Anna Bell Knapp, daughter of John Knapp a shipbuilder of Brooklyn, N. Y. He is a member of the National Republican Club, Knickerbocker (Knickerbocker) Tennis League (Knickerbocker), Country (Knickerbocker), Columbia (Knickerbocker).









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## Publisher's News Letter

The success of the German cross country race will give rise to much discussion as to whether or not this is not the ultimate form that all air races will take. The new record around the world was covered several cross country courses so that a large number of different cities could participate in the affair. If a similar race could be held with Chicago as the hub a very interesting plot could be worked out. One lap could be flown to the Twin Cities and back, another could be to Detroit and Cleveland with a return flight over Dayton, Cincinnati and Indianapolis. A third might go to St. Louis and Kansas City with a return route via Des Moines. The main objective of the routes that could be included brings out the great advantages of this kind of an aerial contest. The day of the enclosed air meet is fast drawing to a close. That was a good way of testing aircraft back in 1911 but with the success in reliability of airplanes, the need for large crowds becomes of increasing importance. Races held in congested locations so as to attract crowds have not been as great factors as they should have been in developing reliable aircraft. Detroit is planning a circuit this year and this means that it will be the start of a new series of cross country races that will interest whole sections rather than single cities. If the N.A.A. would schedule several such meets for different sections of the country, there would probably be greater interest shown in holding meets by location than do not feel like having to finance the grandstands of air races to the extent of several hundred thousand dollars.

Held away in an obscure part of the newspaper was an important item of news that probably escaped many readers. It concerned that Italy had united to America, Navy and Air Service under a single head. This plan which has been in the process of making for the last year is one that will be watched by other countries with the greatest interest. In Great Britain, there has been much discussion of a similar move, for several years, but the attached military and naval services have opposed it so strongly that progress has only been made to the extent of having an Imperial Council of Defense. At one time, it was thought that President Harding was to make an offer toward a unification of the land, sea and air forces of the United States but nothing was ever made public regarding the plan. The virtual dictatorship of Mussolini has made such a change possible in

Italy. While it was objected to by the allied officers, it had the support of those who wished to eliminate duplications and avoid demoralization of command on a scale of efficiency that had to be unobtainable in the last War.

That the experiment Italy is making will be watched closely by the United States goes without saying. It has all the merit of being modern business principles and as a measure of economy it will make a strong appeal. In time of war there has usually been one branch of the national defense that has been the most active. Either the war has been land or aerial, or one of these two forms of warfare has been of the greatest importance. In the future, the air will have to be taken into consideration. It is entirely possible that the air may play a deciding role. This possibility has been questioned by very high authorities, but only time will tell whether they have been short sighted or not. As wars are more and more becoming a matter of national, the great need for a supreme director of supply is causing the attention of all nations to be given greater attention. When those who believe in cross forces of an independent air force are asked for their views on this question, the reply is generally in support of some such arrangement. How the United Air Force will function under the new unified department will be watched by all nations, with the greatest interest.

The great credit cannot be given to our neighboring Air Force over the border. It has been announced that last year the fleet of the Canadian Air Force flew 3340 hours, without a fatal accident. As most of this flying was done in connection with carrying mail, practicing formation and maneuvers, and map making, the record is doubly amazing. So little is heard of the activities of our northern air friends that it is sometimes overlooked that Canada has an independent air force that operates practically independently of the Royal Air Force. While most of its equipment is second hand English manufacturing, there is a possibility that Canada may become a market for specialized planes in this country. In fact, last year some American planes were sold there for special use. Every American War will wish that the record that was made last year will be repeated this year. L.D.G.



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